

FORM - 21B DESIGN CRITERIA FORM FOR CARBON-CARBON AIRCRAFT BRAKE DISCS

In accordance with IMTAR-21, Subpart C3, 21. C3.1.9, 21.C3. 1.10

Reference	
Applicant's Reference	Date:

Material

Project : Name of Inspection Agency : Name of the Developing/Manufacturing Agency:

Manufacturing Process Route : Supply Condition : :

Nominal Dimensions

$\begin{vmatrix} SI \\ No \end{vmatrix}$ or	Comp-	onent cation	Material Property Requirements	Service Conditions	Enviro- nmental and Temp Conditions	Requirements of Metallic Attachments, if any	Any other inform- ation
	onent Name		Physical, Mechanical, Thermal, Friction and Wear				
		Critical/ Non critical	e.g., Physical Properties: Density, Porosity, etc. Mechanical Properties: Flexural Strength. Tensile Strength, Compressive Strength, Shear Strength, Interlaminar Shear Strength, etc. Thermal Properties Specific Heat, Thermal Diffusivity, Thermal Conductivity, CTE, Weight Loss by TGA (Bare + with Anti- Oxidant Coating), etc. Friction and Wear Properties on Brake Dynamometer Co-efficient of Friction, Wear (Thickness Loss), etc.	Kinetic Energy (Normal, Overload, RTO), Brake Application Speed, Brake Pressure, Stop Time, Static Torque, Temperature Rise, Wear (Thickness Loss) etc.			

Enclosures:

- 1. Brief write-up about the Project
- 2. End use of the Components along with justification for classification
- 3. Drawings, photographs of components
- 4. QA Plan

Signature Name & Designation Name of Organisation (with Seal)

Countersigned by CEMILAC